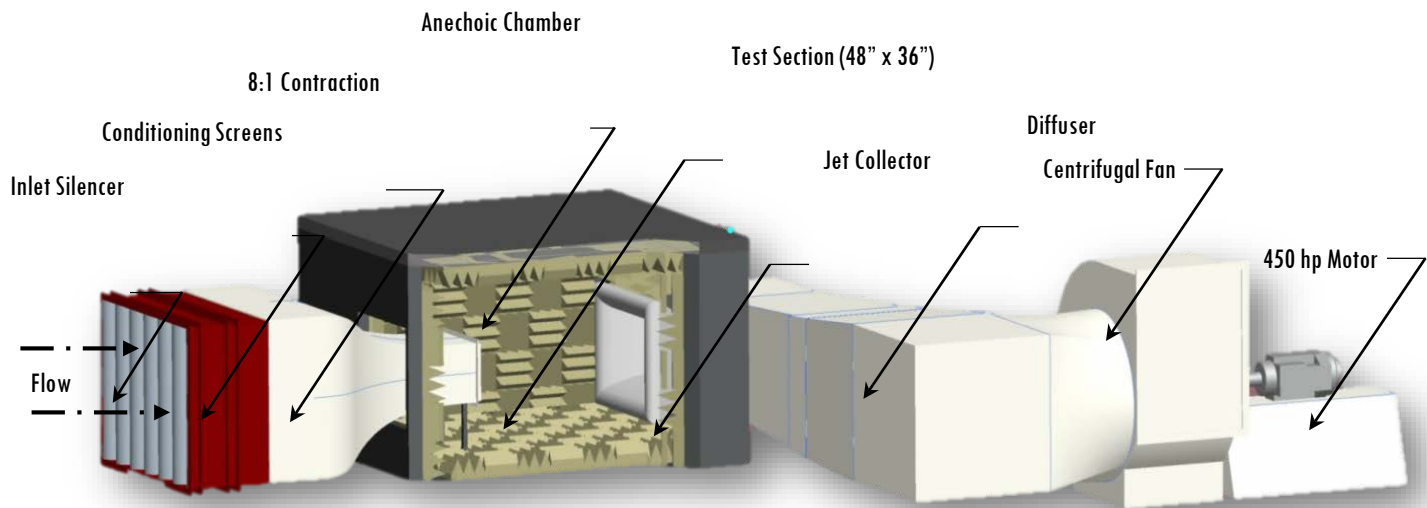




Anechoic Wind Tunnel



Advanced Diagnostics

128-channel Steady Pressure Measurements, Simultaneous

96-channel 24-bit A/D @ 204.8 kHz, Simultaneous

3-axis Force Balance for Lift/Drag Measurements

Hot Wire Anemometry

Particle Image Velocimetry (PIV)

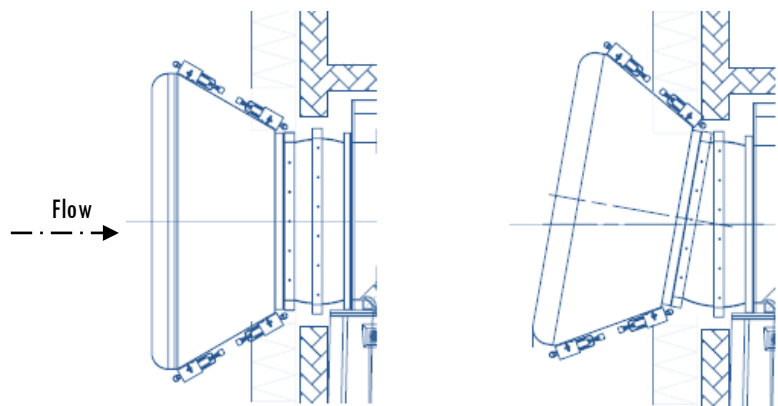
- Stereoscopic
- Tomographic

Pressure Sensitive Paint (PSP)

Laser Doppler Velocimetry (LDV)

Acoustics — Directivity and Source Localization

- Linear Arrays
- 80-Channel Phased Array



Jet collector (top view)

Capable of swiveling $\pm 10^\circ$ off streamwise axis to catch tunnel jet, especially useful for high-lift configurations.

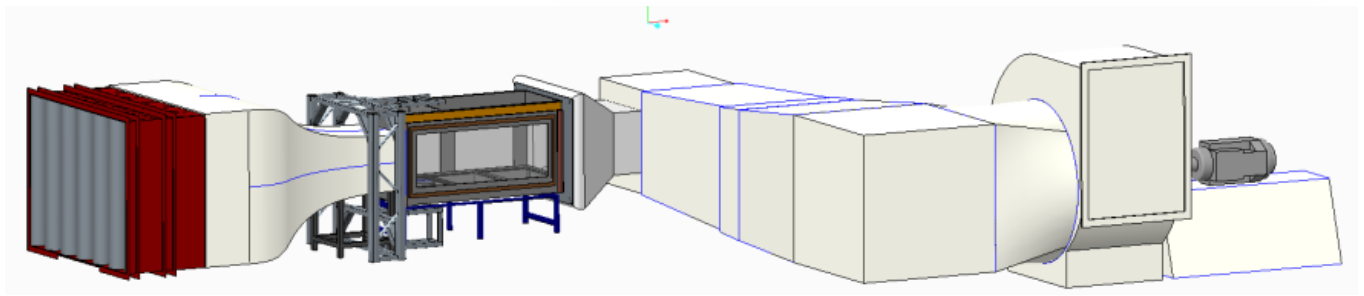
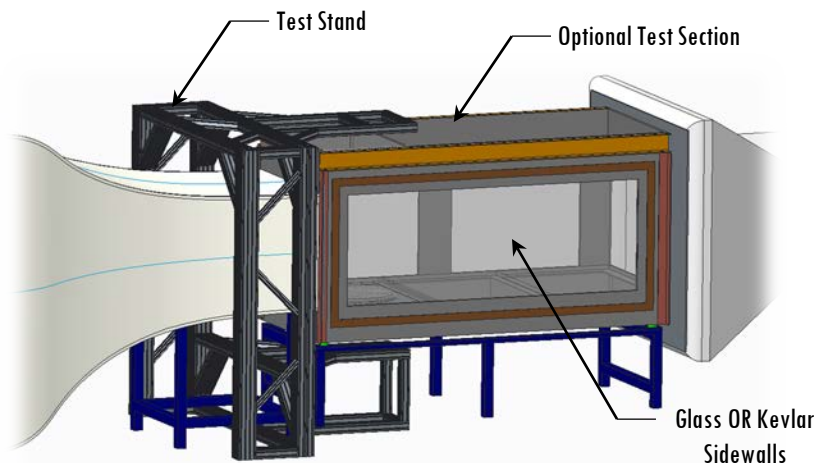
The FSU Aeroacoustic Tunnel (FSAT) is an open-circuit subsonic facility which has a 36" x 48" x 120" test section with a speed range of approximately 18-75 m/s. The facility employs a 450-hp centrifugal fan and has an anechoic chamber with a 250 Hz cut-on frequency enclosing the test section. The test section has three configuration options: open-jet, hard walls, or Kevlar walls, distinguishing itself through versatility. The facility will be used for aerodynamic and aeroacoustic studies of various flow-induced noise phenomenon and boasts state-of-the-art experimental fluid dynamic and aeroacoustic measurement capabilities.

Closed Wall/Kevlar Wall Test Section Configurations

Optional closed walled test section can be outfitted with either hard walls or Kevlar sidewalls.

Hard wall — Glass sidewalls to simulate a conventional aerodynamic tunnel, with optical access for non-intrusive flow measurements.

Kevlar wall — Thin tensioned cloth to contain tunnel jet, acoustically transparent for far field acoustic measurements, Type 120.



Specifications

Item	Specification
Wind Tunnel	Open-Circuit, Open-Jet/Hard Wall/Kevlar Wall
Test Section	48" x 36" x 120"
Model Size and Aspect Ratio, Typ.	0.5 m chord, AR = 2
Mach Number Range	0.05 – 0.22
Anechoic Chamber Low Frequency Cutoff	250 Hz
Background Noise	80 dBA (M = 0.17)
Turbulence Intensity	< 0.1% above 10 Hz
RMS Inlet Flow Nonuniformity	< 1%

Contact

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